

## TEMPORARY REVISION

### TR-OÄM-20-247

### Aluminium MLG

This Temporary Revision TR-OÄM-20-247 is approved in conjunction with the Optional Design Change Advisory OÄM 20-247/a and is valid in conjunction with the latest revision of the DV 20-100 Airplane Flight Manual until this temporary revision has been incorporated into the Airplane Flight Manual.

The limitations and information contained herein either supplement or, in the case of conflict, override those in the Airplane Flight Manual.

The technical information contained in this document has been approved under the authority of DOA No. EASA.21J.052.

Doc. No.	Section	Affected Pages
4.01.20-E	1	1-5a
	7	7-6a

#### Instruction:

- Print this document on yellow paper (single-sided).
- Insert this cover page as the first page of the AFM.
- Insert the other pages of this Temporary Revision in front of the corresponding AFM pages.

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**Affected Chapters:**

**1.5 DIMENSIONS**

*The following is added:*

**LANDING GEAR**

Main wheel 380\*150/15\*6.00-5

**I** If the aluminium MLG is installed: 5.00-5, 6 ply

## 7.5 LANDING GEAR SYSTEM

*The first paragraph is amended to read:*

The landing gear system consists of the two main landing gear wheels mounted to a self-spring steel strut (an aluminum strut, if TR-OÄM-20-247 is installed) and a free following nose wheel. The suspension of the nose wheel is handled by an elastomer package.