

SERVICE INFORMATION NO. SI-42-121

NOTE: SI's are used only:

1) To distribute information from DAI to our customers.

2) To distribute applicable information/documents from our suppliers to our customers with additional information.

Typically there is no revision service for SI's. Each new information or change of that will be sent along with a new SI.

I. TECHNICAL DETAILS

1.1 Airplanes affected:

All DA 42 and DA 42 M aircraft equipped with Thielert TAE 125-02-99 engines.

1.2 Subject:

EASA Airworthiness Directive No. 2009-0224 ATA-Code: 61

1.3 Reason:

EASA has issued the Airworthiness Directive No. 2009-0224 which mandates the installation of the Vibration Isolator P/N 05-7212-K022302 and the Proportional Pressure Reducing Valve P/N 05-7212-E002801 in accordance with TM TAE 125-1009 P1 Revision 3 and TM TAE 125-1007 P1 Revision 2.

1.4 Information:

For detailed technical information refer to EASA AD No. 2009-0224, which is applicable without any further additions or restrictions.

Note: This AD supersedes AD 2009-0151, retaining the requirements and reducing the compliance time from 110 to 55 flight hours.

II. OTHERS

The EASA AD No. 2009-0224 is attached to this SI.

In case of doubt contact Thielert Aircraft Engines GmbH.

EASA AIRWORTHINESS DIRECTIVE AD No.: 2009-0224 Date: 20 October 2009 Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation. This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption]. Type Approval Holder's Name : Type/Model designation(s) : Thielert Aircraft Engines GmbH TAE125-02-99 engines TCDS Number : EASA.E.055 Foreign AD : Not applicable Supersedure: This AD supersedes EASA AD 2009-0151, dated 10 July 2009. Engine/Propeller – Constant Speed Unit – Propeller Control **ATA 61** Valve – Vibration Isolator - Installation Manufacturer(s): Thielert Aircraft Engines GmbH. TAE125-02-99 engines, all serial numbers, if installed on Diamond Aircraft Applicability: Industries Model DA 42 and DA 42 M aircraft. Reason: In-flight engine shutdown incidents have been reported on Diamond Aircraft Industries DA 42 aircraft equipped with TAE125 engines. The investigations showed that it was mainly the result of failure of the Proportional Pressure Reducing Valve (Propeller Control Valve) due to high vibration. This condition, if not corrected, could lead to further cases of engine in-flight shutdown, possibly resulting in reduced control of the aircraft. As this problem has only manifested itself on these engines installed on Diamond Aircraft Industries DA 42 aircraft, TAE125-02-99 engines installed on other aircraft are not affected. In addition to the information stated above, it has been found out that failure of the electrical connection to the Proportional Pressure Reducing Valve 24V (Propeller Control Valve) Part Numbers (P/N) NM-0000-0124501 and 05-7212-K021401 contributed to power loss events or IFSD. To address and correct this situation, Thielert Aircraft Engines GmbH has published TM TAE 125-1009 P1, providing instructions for installation of the Vibration Isolator P/N 05-7212-K022302 that reduces the vibration

level on the Proportional Pressure Reducing Valve. For preventing failures

	of the valve's electrical connection, a new Proportional Pressure Reduci Valve 24V (Propeller Control Valve) Part Number (P/N) 05-7212-E00280 has been introduced by TM TAE 125-1007 P1 Revision 2.		
	For the reasons described above, EASA AD 2009-0151 required the installation of the Vibration Isolator and the new Proportional Pressure Reducing Valve 24V (Propeller Control Valve). This AD supersedes AD 2009-0151, retaining the requirements and reducing the compliance time from 110 to 55 flight hours.		
Effective Date:	03 November 2009		
Required Action(s) and Compliance Time(s):	 Required as indicated unless already accomplished: Within 55 flight hours from the effective date of this AD, at the next inspection or until 31 January 2010, whichever occurs first, install the Proportional Pressure Reducing Valve 24V (propeller control valve) Part Number (P/N) 05-7212-E002801 and the Vibration Isolator P/N 05-7212-K022302 in accordance with TM TAE 125-1007 P1 Rev. 2 and TM TAE 125-1009 P1 Revision 3. Installations done in accordance with previous issues of TM TAE 125-1009 P1 are acceptable for compliance with the requirements of this AD. Note: Vibration Isolator and Proportional Pressure Reducing Valve (Propeller Control Valve) are part of the Gearbox assembly. The Gearbox, 		
	and thus Vibration Isolator and Proportional Pressure Reducing Valve, has to be inspected regularly in accordance with OM-02-02 and RM-02-02.		
Ref. Publications:	 Thielert Service Bulletin TM TAE125-1009 P1 Rev. 3 dated 14 October 2009. Thielert Service Bulletin TM TAE125-1007 P1 Rev. 2 dated 29 April 2009. Thielert Service Bulletin TM TAE125-1009 P1 dated 07 July 2009. Engine Operation & Maintenance Manual OM-02-02 Issue 1 Rev. 7 dated 17 June 2009. Engine Repair Manual RM-02-02 Issue 1 Rev. 15 dated 29 April 2009. The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD. 		
Remarks :	 If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD. The required actions and the risk allowance have granted the issuance of a Final AD with Request for Comments, postponing the public consultation process after publication. Enquiries regarding this AD should be referred to the Airworthiness Directives, Safety Management & Research Section, Certification Directorate, EASA. E-mail <u>ADs@easa.europa.eu.</u> For any question concerning the technical content of the requirements in this AD, please contact: Thielert Aircraft Engines Platanenstraße 14 D-09350 Lichtenstein, Germany Telephone +49-37204-696-0; Fax +49-37204-696-55; E-mail info@centurion-engines.com 		

EASA	AIRWORTHINESS DIRECTIVE			
	AD No.: 2009- 015	51		
	Date: 10 July 2009			
	Regulation (EC) No 216/2	Directive (AD) is issued by EASA, acting in accordance with 2008 on behalf of the European Community, its Member States I countries that participate in the activities of EASA under Article		
This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].				
Type Approval Holder's Name :		Type/Model designation(s) :		
Thielert Aircraft Engines GmbH		TAE125-02-99 engines		
TCDS Number : EASA.E.055				
Foreign AD : Not applicable				
Supersedure: None				
ATA 61		Constant Speed Unit – Propeller Control Valve – Vibration Isolator		
Manufacturer(s):	Thielert Aircraft Engine	s GmbH.		
Applicability:	TAE125-02-99 engines, all serial numbers, if installed on Diamond Aircraft Industries Model DA 42 and DA 42 M aircraft.			
Reason:	In-flight engine shutdown incidents have been reported on Diamond Aircraft Industries DA 42 aircraft equipped with TAE125 engines. The investigations showed that it was mainly the result of failure of the Proportional Pressure Reducing Valve (Propeller Control Valve) due to high vibration. This condition, if not corrected, could lead to further cases of engine in-flight shutdown, possibly resulting in reduced control of the aircraft.			
		nly manifested itself on these engines installed on tries DA 42 aircraft, TAE125-02-99 engines aft are not affected.		
	failure of the electrical of Valve 24V (Propeller C	nation stated above, it has been found out that connection to the Proportional Pressure Reducing ontrol Valve) Part Numbers (P/N) NM-0000- K021401 contributed to power loss events or IFSD.		
	has published TM TAE	this situation, Thielert Aircraft Engines GmbH 125-1009 P1, providing instructions for installation P/N 05-7212-K022302 that reduces the vibration		

	level on the Proportional Pressure Reducing Valve. For preventing failures of the valve's electrical connection, a new Proportional Pressure Reducing Valve 24V (Propeller Control Valve) Part Number (P/N) 05-7212-E002801 has been introduced by TM TAE 125-1007 P1 Revision 2.		
	For the reasons described above, this EASA AD requires the installation of the Vibration Isolator and the new Proportional Pressure Reducing Valve 24V (Propeller Control Valve).		
	The content of this AD was partially addressed by PAD 09-071. One comment was received and is answered in the CRD 09-071, which can be found at http://ad.easa.europa.eu/ .		
Effective Date:	24 July 2009		
Required Action(s) and Compliance Time(s):	Required as indicated unless already accomplished:		
	Within 110 flight hours from the effective date of this AD, install the Proportional Pressure Reducing Valve 24V (propeller control valve) Part Number (P/N) 05-7212-E002801 and the Vibration Isolator P/N 05-7212-K022302 in accordance with TM TAE 125-1007 P1 Rev. 2 and TM TAE 125-1009 P1.		
	Note: Vibration Isolator and Proportional Pressure Reducing Valve (Propeller Control Valve) are part of the Gearbox assembly. The Gearbox, and thus Vibration Isolator and Proportional Pressure Reducing Valve, has to be inspected regularly in accordance with OM-02-02 and RM-02-02.		
Ref. Publications:	Thielert Service Bulletin TM TAE125-1007 P1 Rev. 2 dated 29 April 2009.		
	Thielert Service Bulletin TM TAE125-1009 P1 dated 07 July 2009.		
	Engine Operation & Maintenance Manual OM-02-02 Issue 1 Rev. 7 dated 17 June 2009.		
	Engine Repair Manual RM-02-02 Issue 1 Rev. 15 dated 29 April 2009.		
	The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD.		
Remarks :	1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.		
	2. The required actions and the risk allowance have granted the issuance of a Final AD with Request for Comments, postponing the public consultation process after publication.		
	 Enquiries regarding this AD should be referred to the Airworthiness Directives, Safety Management & Research Section, Certification Directorate, EASA. E-mail <u>ADs@easa.europa.eu.</u> 		
5	 For any question concerning the technical content of the requirements in this AD, please contact: 		
	Thielert Aircraft Engines Platanenstraße 14 D-09350 Lichtenstein, Germany Telephone +49-37204-696-0; Fax +49-37204-696-55; E-mail <u>info@centurion-engines.com</u>		