

DAI MSB 42-143/1 DAI MSB 42NG-086/1 Page 1 of 4 25-Jan-2022

MANDATORY SERVICE BULLETIN MSB 42-143/1 MSB 42NG-086/1

supersedes MSB 42-143 / MSB 42NG-086

I TECHNICAL DETAILS

I.1 Category

Mandatory.

I.2 Airplanes affected

Type: DA 42, DA 42 M, DA 42 NG, DA 42 M-NG Serial numbers: 42.004 through 42.427, 42.AC001 through 42.AC152, 42.M001 through 42.M026, 42.N001 through 42.N067, 42.N100 through 42.N265, 42.N300 through 42.N423, 42.N427 through 42.N450, 42.MN001 through 42.MN034, 42.MN037 through 42MN042, 42.MN050 through 42.MN055, 42.MN057, 42.MN058, 42.MN100 through 42.MN112, 42.NC001 through 42.NC004, 42.NC006 through 42.NC008, 42.NW001 through 42.NW090, 42.NW093 through 42.NW095, 42.MNW001 through 42.MNW004

I.3 Date of effectivity

23-Dec-2021

I.4 <u>Time of Compliance</u>

Within the next 200 flight hours from the date of effectivity, but not later than 31-Dec-2022.

I.5 Subject

Inspection of hole position and condition in the rudder steering bracket.

ATA-Code: 57-10



I.6 <u>Reason</u>

Reports of worn holes in the rudder steering bracket were received, that were caused by incorrectly tightened self-locking nuts.

This condition, if not corrected, may lead to loss of the primary directional control system.

This Service Bulletin provides instructions to identify the condition of the hole and to measure the distance between the hole and the edge of the rudder steering bracket.

After completion of this Service Bulletin the repetitive inspection of the torque of the T-yoke bolt is replaced by visual inspection of torque seal marks.

I.7 Concurrent Documents

None.

I.8 <u>Approval</u>

The technical information or instructions contained in this document relate to the Design Change Advisory No. MÄM 42-1213/a, which has been approved under the authority of EASA Design Organization Approval ref. EASA.21J.052.

The technical content of this document has been approved under the authority of DOA ref. EASA.21J.052.

I.9 Accomplishments / Instructions

Comply with WI-MSB 42-143 / WI-MSB 42NG-086, latest effective issue.

I.10 Mass (Weight) and CG

No change.

II PLANNING INFORMATION

II.1 Material and Availability

See WI-MSB 42-143 / WI-MSB 42NG-086, latest effective issue.

II.2 Special Tools

None.

II.3 Labour Effort

Approx. 1.5 hours when done along with scheduled 200 hour or annual inspection, approx. 4.5 hours otherwise.

II.4 Credit

For all airplanes within warranty period, if performed in conjunction with an annual inspection.



II.5 <u>Reference Documents</u>

DA 42 Series Airplane Maintenance Manual, Doc. No. 7.02.01, latest effective issue.

DA 42 NG Airplane Maintenance Manual, Doc. No. 7.02.15, latest effective issue.

III <u>REMARKS</u>

- 1. All work must be done by a certified aircraft service station or a certified aircraft maintenance mechanic.
- 2. All work, in particular that which is not especially described in this Service Bulletin, must be done in accordance with the referenced Maintenance Manual.
- 3. Completion of all work must be recorded in the log book.
- 4. If material and/or labor hours are subject to be credited through Diamond Aircraft Industries, the Service Bulletin must be done by an authorized Diamond Service Center within the time of compliance and the Warranty Application incl. Work Report must be sent not later than 30 days after completion of work.
- 5. In case of doubt contact Diamond Aircraft Industries GmbH.



EXECUTION REPORT TO SERVICE BULLETIN MSB 42-143/1 MSB 42NG-086/1

AIRPLANE INFORMATION				
Airplane Serial Number				
Airplane Registration				
Airplane Operator				
Hours of operation of airplane				
No. of landings				
Hours of operation-engine	LH .			
	RH .			
Typical operation of airplane	private, clu	ıb, training	, other	
Oblong hole washer found to be installed	b		yes / no	
Distance d between edge of hole and ec	lge of rudde	er steering	bracket.	
Replacement of rudder steering bracket			yes / no	
Worn hole found			yes – dimensions / no	
Repair of rudder steering bracket necess	sary (bushin	ng)	yes / no	
Accomplishment with RSB 42-139 / RSB 42NG-081			yes – date / no	
Salesforce case number, if known				

Date, Name, Sign

Please send the completed form to executionreports@diamondaircraft.com



WORK INSTRUCTION WI-MSB 42-143 WI-MSB 42NG-086

I GENERAL INFORMATION

I.1 Subject

Inspection of hole position and condition in the rudder steering bracket.

I.2 <u>Reference Documents</u>

DA 42 Series Airplane Maintenance Manual, Doc. No. 7.02.01, latest effective issue.

DA 42 NG Airplane Maintenance Manual, Doc. No. 7.02.15, latest effective issue.

I.3 <u>Remarks</u>

- a) All work must be done by a certified aircraft service station or a certified aircraft maintenance mechanic.
- b) All work, in particular if not described in this work instruction, must be done in accordance with the referenced maintenance manual.
- c) For conversion factors between SI units and US/Imperial units refer to AMM Chapter 02.
- d) In case of doubt, contact Diamond Aircraft Industries GmbH.

II DRAWINGS, SPECIAL TOOLS & MATERIALS

II.1 Drawings

None.

II.2 Special Tools

None.



II.3 <u>Material</u>

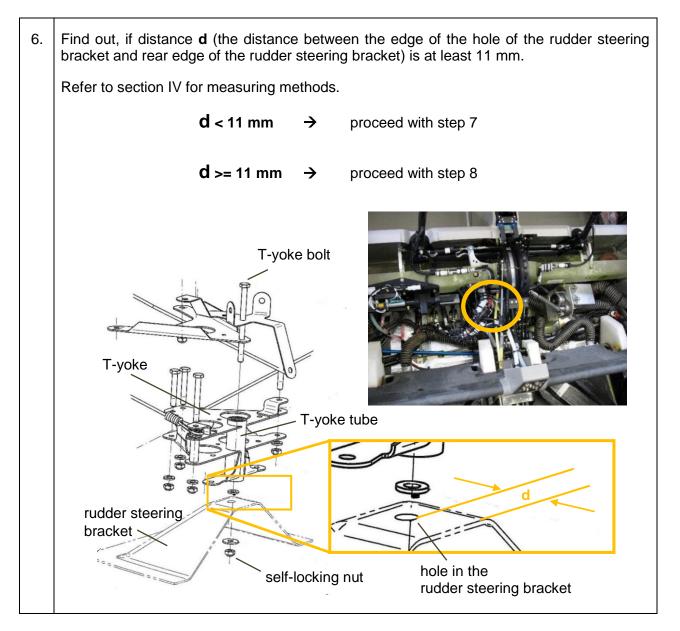
Quantity	Part Number	Description
3	DIN-985-M6-A2	Self-Locking Nut
2	MS 21256-1	turnbuckle clip
as required	D60-5320-00-32	T-yoke bolt

Material is available from Diamond Aircraft Industries.

III INSTRUCTIONS

1.	Insert AMM-TR-MÄM-42-1213 into the AMM or use an AMM revision, in which AMM-TR-MÄM-42-1213 has been incorporated.
2.	Remove the pilot and co-pilot seat in accordance with AMM Section 25-10.
3.	Remove the middle tunnel front cover in accordance with AMM Section 25-10.
4.	 Remove components for better access to the rudder steering bracket: Remove the two heating hoses, that are indicated in the picture by the arrows, from the aluminum Y-connector, pull out the heating hose, that is routed along the rudder steering bracket, disconnect the short aileron push rod from the main spar lever as indicated by the circle.
5.	Disconnect the NLG steering rod at the NLG steering cardan in the wheel well.







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7.	If distance d is less than 11mm:		
	 disconnect the NLG steering rod from the T-yoke, 		
	 disconnect all rudder control cables from the T-yoke, 		
	remove the T-yoke bolt,		
	remove the T-yoke,		
	 make sure that all washers are removed from the airplane, 		
	 confirm that the oblong hole washer is or is not installed, 		
	 confirm orientation of the oblong hole washer, if installed, 		
	 confirm dimension of distance d by measuring the distance between the edge of the hole and the edge of the bracket on rudder steering bracket, 		
	 send all information to <u>support-austria@diamondaircraft.com</u>, containing: 		
	 detailed photographs of the hole together with a measuring tape, positioned beside the hole, 		
	 the filled-in <u>damage report</u>, which can be downloaded from DAI technical publications home page 		
	 request repair instructions for the rudder steering bracket, 		
	 proceed with step 14 after repair. 		
8.	Remove the passenger seat in accordance with AMM, Section 25-10.		
9.	Release rudder cable turnbuckles.		
10.	Remove self-locking nut of T-yoke bolt:		
	Remove the Fokker Needle, if installed,		
	 block the self-locking nut with a ring-wrench, 		
	 turn the T-yoke bolt until the self-locking nut falls off, 		
	 remove washer and self-locking nut from the airplane, 		
	discard the self-locking nut.		
	Note: Make sure the washer between rudder steering bracket and T-yoke does not get lost, if oblong hole washer D60-5320-00-31 or no oblong hole washer is installed.		



- 11. Examine the hole in the rudder steering bracket for wear and play. Look especially for possible lateral movement of the lower part of the T-yoke:
 - Apply simultaneous rudder pedal force to the RH pilot rudder pedal and LH co-pilot rudder pedal.
 - Apply rudder pedal force and move the rudder from full LH deflection to full RH deflection.
 - Apply hand force to the T-yoke and tilt it back / forth and sideward.
 - Use an inspection mirror to examine the hole in the bottom side of the rudder steering bracket.



If a worn hole has been found, proceed with step 12.

If the hole is not damaged, proceed with step 13.

- 12. If a worn or enlarged hole, or play of the T-yoke bolt in the hole of the rudder steering bracket has been found,
 - disconnect the NLG steering rod from the T-yoke,
 - disconnect all rudder control cables from the T-yoke,
 - remove the T-yoke bolt,
 - remove the T-yoke,
 - make sure that all washers are removed from the airplane,
 - send all information to support-austria@diamondaircraft.com, containing:
 - dimensions of the worn / enlarged hole in the rudder steering bracket
 - detailed photographs of the worn / enlarged hole together with a measuring tape, positioned beside the hole,
 - filled-in <u>damage report</u>, which can be downloaded from DAI technical publications home page
 - request instructions for repair of the hole in the rudder steering bracket,
 - proceed with step 14 after repair.



13.		install T-yoke bolt D60-5320-00-32 with additional hole for a Fokker Needle, with MSB 42-146 / MSB 42NG-087.
14.	Install a new se	elf-locking nut to the T-yoke bolt.
	 Turn the addition Remove Turn the T-yoke Install F Turn the Close the Block the Torque Remove Turn the pointy endowing the second secon	 a self-locking nut with an open-end wrench. a T-yoke bolt, until the self-locking nut is positioned above the hole for the lal Fokker Needle. Do not yet tighten the T-yoke bolt. b the open-end wrench. c T-yoke bolt to a position, so that the Fokker Needle can be installed to the bolt. cokker Needle. c T-yoke bolt by 180°. he Fokker Needle. ne self-locking nut with an open-end wrench. the T-yoke bolt with 6 Nm. the open-end wrench. c T-yoke bolt without blocking the self-locking nut to a position, so that the self of the rudder steering (refer to sketch).
	TorqueMake su	te self-locking nut with an open-end wrench. the T-yoke bolt with 9 Nm. ure the pointy end of the Fokker Needle is hidden under the rudder steering and does not damage the heating hose.
	CAUTION:	Tighten the T-yoke bolt to the correct torque <u>only</u> with released rudder cable turnbuckles.
	CAUTION:	It is essential to use a torque wrench, in order assure application of the correct torque and to avoid inadvertent loosening of the self-locking nut.
	CAUTION:	If the T-yoke bolt is already tightened to the correct torque, do not change the orientation of the Fokker Needle by turning the T-yoke bolt.
	Note:	The Fokker Needle can damage the heating hose if the pointy end is not hidden under the rudder steering bracket.



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15.	Mark the head of the T-yoke bolt and the self-locking nut with at least two torque seal marks.
	<image/>
16.	Adjust left and right rudder cable turnbuckles to give the correct tension to the control cables in accordance with AMM, Section 27-20.
17.	Install all heating hoses.
18.	Connect the short aileron push rod to the idler lever at the main spar.
19.	Connect the NLG steering rod to the NLG steering cardan in the wheel well.
20.	Install the passenger seat in accordance with AMM, Section 25-10.
21.	Install the middle tunnel front cover.
22.	Install the pilot and copilot seat in accordance with AMM Section 25-10.
23.	Clean working areas, check for foreign objects.
24.	Check all altered, replaced, repaired parts for proper function.
25.	Test all systems in working area for function.
26.	Make all necessary entries in the airplane logs.
27.	Fill in the execution report and send it to execution reports@diamondaircraft.com



IV Measurement of distance d

This section provides methods to find out, if distance **d** is at least 11 mm.

IV.1 Direct measurement

1.	Determine distance d by direct measurement with the T-yoke removed.		
	Note: When measuring distance d directly, consider the orientation of the oblong hole washer, if installed. Refer to IV.2 , step 6.		
	Distance d =		
	Return to section III.		

IV.2 Indirect measurement

1.	Use a woo determine o	den strip (cross-section e.g. 15 mm x 3 mm) and a piece of modeling clay to distance d .
	Put	a small piece of modeling clay to the end of a wooden strip.
	• Hold	d the wooden strip to the metal tube of the rudder bell crank.
		ve the wooden strip down, so that the modeling clay is pressed onto the rudder ering bracket.
	Note:	Make sure there is no modeling clay between the wooden strip and the metal tube of the rudder bell crank.
	Note:	For illustration, this step is shown on an airplane with removed components in the area of the rudder steering bracket.
		<image/>



2.	Measure distance x .
3.	If distance \mathbf{x} is less than 3 mm (.12 in), distance \mathbf{d} is less than 11 mm. Return to section III.
	If distance x is greater than or equal to 6 mm (.24 in), distance d is at least 11 mm. Return to section III.
	If distance \mathbf{x} is between 3 mm (.12 in) and 6 mm (.24 in), proceed with the following steps.
N	lote: As an alternative method an appropriate material (stick) with a defined thickness of 3 mm and 6 mm can be used, to check if distance x is less than 3 mm or if distance x is greater than or equal to 6 mm.
4.	Check, if an oblong hole washer has been installed to the rudder steering bracket in accordance with MSB 42-087 / MSB 42NG-016:
	Obtain information from the maintenance logs,
	 Slightly lift off the T-yoke and use an inspection mirror to check, if an oblong hole washer is installed to the rudder steering bracket.
	CAUTION: Make sure, the washer between the T-yoke and the rudder steering bracket does not get lost.



