

RECOMMENDED SERVICE BULLETIN

RSB 42-125

RSB 42NG-065

I TECHNICAL DETAILS

I.1 Category

Recommended.

I.2 Airplanes affected

Type: DA 42 /M /NG /M-NG
Serial numbers: 42.055, 42.058,
42.065 through 42.106,
42.108, 42.114 through 42.176
42.178 through 427
42.AC001 through 42.AC151
42.M001 through 42.M026
42.N001 through 42.N067
42.N100 through 42.N217, 42.N219
42.NC001 through 42.NC008
42.MN001 through 42.MN040
42.MN050 through 42.MN052

All aircraft to which a replacement passenger door or replacement canopy has been installed that has an aluminum door locking mechanism having P/N D41-5211-60-00_2 which is installed to canopies having P/Ns D60-5210-00-00, D60-5210-00-00X02, D60-5210-00-00X03 or D62-5210-00-00 and passenger doors having P/Ns D60-5221-00-00_1 or D60-5221-00-00_2.

I.3 Date of effectivity

19-Jun-2017

I.4 Time of Compliance

Within 200 flight hours or within 1 year from the date of effectivity, whichever is reached first.

I.5 Subject

Replacement of steel pin of door handle mechanism of canopy and passenger door with an improved design.

ATA-Code: 52-00

I.6 Reason

DAI has received 4 reports of failed steel pins that attach the compression gas spring to the door locking mechanism frame.

In order to avoid further steel pin failures Diamond Aircraft Industries recommends replacing the steel pin with a reinforced steel pin design.

I.7 Concurrent Documents

None.

I.8 Approval

The technical information or instructions contained in this document relate to Design Change Advisory No. MÄM 42-934/a, which has been approved under the authority of EASA Design Organization Approval ref. EASA.21J.052.

The technical content of this document has been approved under the authority of DOA ref. EASA.21J.052.

I.9 Accomplishments / Instructions

Comply with WI-RSB 42-125 / WI-RSB 42NG-065, latest effective issue.

I.10 Mass (Weight) and CG

The change in mass and CG is negligible.

II PLANNING INFORMATION**II.1 Material and Availability**

See WI-RSB 42-125 / WI-RSB 42NG-065, latest effective issue.

II.2 Special Tools

None.

II.3 Labour Effort

2 hours per door.

II.4 Credit

For all AC within warranty period.

II.5 Reference Documents

DA 42 Series Airplane Maintenance Manual, Doc. No. 7.02.01, latest effective issue.

DA 42 NG Airplane Maintenance Manual, Doc. No. 7.02.15, latest effective issue.

III REMARKS

1. All work must be done by a certified aircraft service station or a certified aircraft maintenance mechanic.
2. All work, particular that which is not especially described in this Service Bulletin, must be done in accordance with the referenced Maintenance Manual.
3. Completion of all work must be recorded in the log book.
4. If material and/or labor hours are subject to be credited through Diamond Aircraft Industries, the Service Bulletin must be done by an authorized Diamond Service Center within the time of compliance and the Warranty Application incl. Work Report must be sent not later than 30 days after completion of work.
5. In case of doubt contact Diamond Aircraft Industries GmbH.

**EXECUTION REPORT TO
SERVICE BULLETIN
RSB 42-125
RSB 42NG-065**

AIRPLANE INFORMATION

Airplane Serial Number _____

Airplane Registration _____

Airplane Operator _____

Hours of operation of airplane _____

No. of landings _____

Hours of operation-engine _____

Typical operation of airplane private, club, training, other _____

Date, Name, SignPlease e-mail the execution report to executionreports@diamond-air.at

WORK INSTRUCTION

WI-RSB 42-125

WI-RSB 42NG-065

I GENERAL INFORMATION

I.1 Subject

Replacement of steel pin of door handle mechanism of canopy and passenger door with an improved design.

I.2 Reference Documents

DA 42 Series Airplane Maintenance Manual, Doc. No. 7.02.01, latest effective issue.

DA 42 NG Airplane Maintenance Manual, Doc. No. 7.02.15, latest effective issue.

I.3 Remarks

- a) All work must be done by a certified aircraft service station or a certified aircraft maintenance mechanic.
- b) All work, in particular if not described in this work instruction, must be done in accordance with the referenced maintenance manual.
- c) For conversion factors between SI units and US/Imperial units refer to AMM Chapter 02.
- d) In case of doubt, contact Diamond Aircraft Industries GmbH.

II DRAWINGS, SPECIAL TOOLS & MATERIALS

II.1 Drawings

None.

II.2 Special Tools


None.

II.3 Material

Quantity	Part Number	Description
2	D41-5211-60-36_03	steel pin (tie bolt 3)
2	DIN 125A-5.3-PA	washer
2	BU2-08-05-132-N1	distance bush
a/r	D41-5211-60-91	compression gas spring

Material is available from Diamond Aircraft Industries.

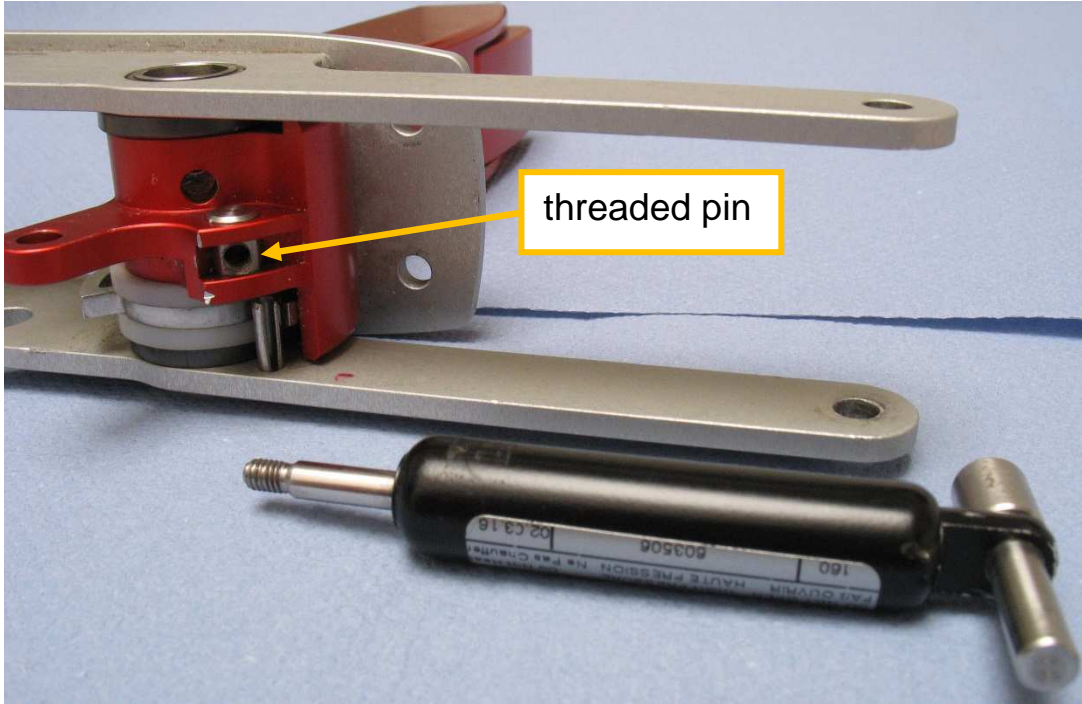
II INSTRUCTIONS

1	Make sure that AMM-TR-MÄM 42-994 has been incorporated into the AMM.
2	Test the compression gas spring of the passenger door in accordance with AMM, section 52-10.
3	Remove the door handle mechanism from the passenger door in accordance with AMM, section 52-10.
4	Remove the 2 circlips as shown in the pictures blow. 
5	Remove the steel pin, the polyamide distance bush and the polyamide washer from the door handle mechanism frame.

6

If necessary, replace the compression gas spring of the passenger door handle mechanism:

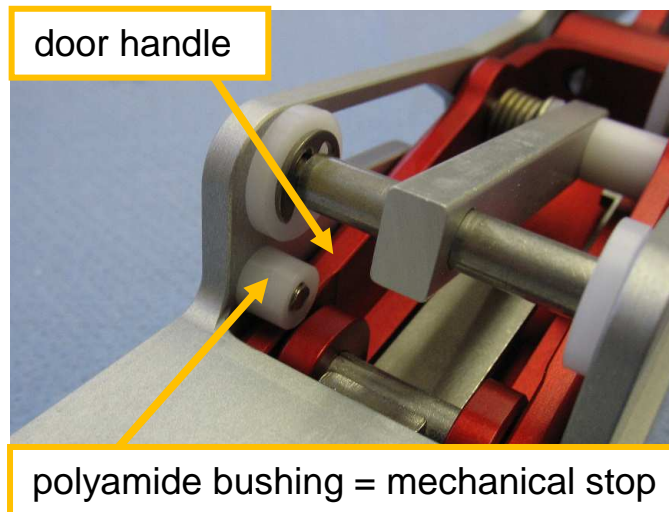
- Unscrew the piston of the compression gas spring from the threaded pin.
- Install the new compression gas spring to the threaded pin.



Note: Secure the thread of the compression gas spring with Loctite 243 when installing the new compression gas spring to the threaded pin.

7

Adjust the length of the compression gas spring. Make sure that the polyamide bushing defines the mechanical stop of the red aluminum door handle as shown in the picture below.

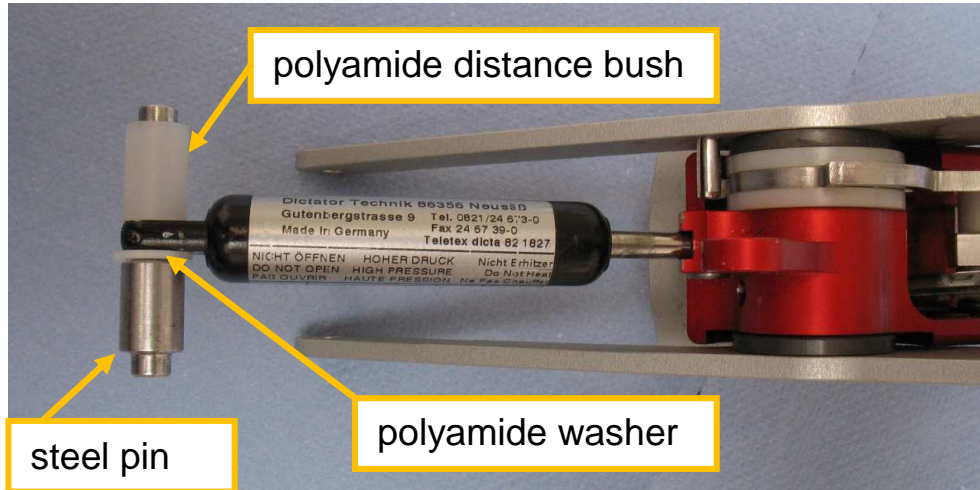


Note: The mechanical stop of the red aluminum door handle must not be defined by the internal stop of the compression gas spring.

8

Prepare the new steel pin for installation into the door handle mechanism frame:

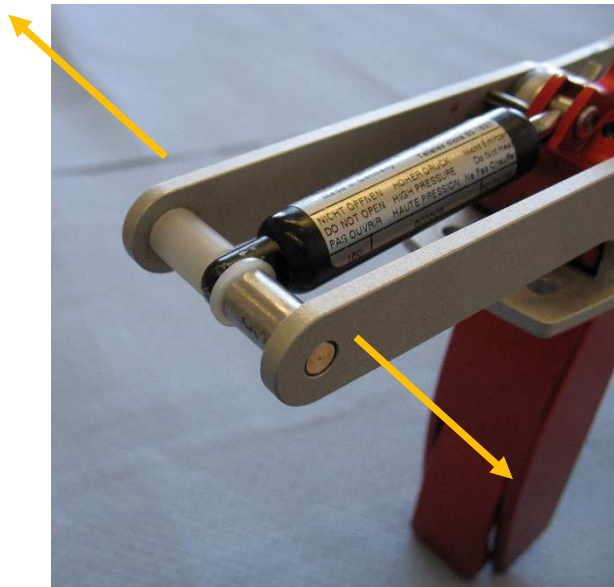
- Move the polyamide washer onto the new steel pin.
- Move the steel pin through the eye end of the compression gas spring.
- Move the polyamide distance bush onto the steel pin.



9

Install the new steel pin with the polyamide washer and the polyamide distance bush to the door handle mechanism frame:

- Move the steel pin into position.
- Carefully bend the side parts of the door handle mechanism frame away from each other as shown by the yellow arrows in the picture below so that the steel pin can snap into the holes of the side parts of the door handle mechanism frame.



Caution: Bend the side parts of the door handle mechanism frame away from each other only as far as necessary. If the side parts are bent too much and permanently deformed, the whole door handle mechanism must be replaced.

10	Install the door handle mechanism to the passenger door in accordance with AMM, section 52-10.
11	Remove the door handle mechanism from the canopy in accordance with AMM, section 52-10.
12	Perform items 4, 5, 7, 8 and 9 on the door handle mechanism of the canopy.
13	Install the door handle mechanism of the canopy in accordance with AMM, section 52-10.
14	Clean working area, check for foreign objects.
15	Check all altered, replaced, repaired parts for proper function.
16	Test all systems in working area for function.
17	Make all necessary entries in the airplane logs.